Complexity out of simplicity in Boltzmann equation and decomposed computation

July 5th Thu, 2018 (12:00~12:30PM)

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> Presented at Phil Roe's 80th Birthday Celebrations Maxwell Centre, Cambridge, UK



Research Center for Aircraft Core Technology



Models in fluid dynamics



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Talk 1/20

Models in magneto-hydrodynamics (MHD)

3*3 quadratic MHD model (1997; Myong & Roe)



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Holistic big picture of fluid dynamics; two extremes

(High Reynolds) fluid dynamics is difficult because:



At extremely small scales, even turbulent flow is very simple. It is smooth and well behaved. At larger scales, however, a fluid is subject to very few constraints. It can develop arbitrary levels of **complexity** like the effect of turbulence on separation. (P. L. Roe, "Future developments in CFD," ICAAT-GNU, May 2014)

(Low Reynolds mesoscopic) fluid dynamics is difficult because:



It involves **microscopic** collisional interactions among fluid particles and their **interplay** with the kinematic motion of particles in the macroscopic framework. This challenge is vividly illustrated by the high Mach number shock singularity problem (HMNP). (R. S. Myong, PoF 2014)



Shift of focus in Burgers and need of kinetic theory



What does **interplay** mean? Between the **kinematic motion** of particles in the macroscopic framework and microscopic **collisional interactions** among fluid particles.

In order to answer this question, one needs to look at the molecular description of gas particles; **gas kinetic theory** or **Boltzmann-based gas dynamics**



Simplicity in Boltzmann transport equation (BTE)

- Assumptions; 1) the mean free path >> the effective range of the intermolecular forces; 2) the velocities of the colliding particles are uncorrelated (molecular chaos and break of time reversal)
- A first-order partial differential equation with an integral collision term

 $\begin{pmatrix} \frac{\partial}{\partial t} + \mathbf{v} \cdot \nabla \end{pmatrix} f(t, \mathbf{r}, \mathbf{v}) = \frac{1}{Kn} C[f, f_2] \quad \text{Only two effects! (Simplicity)} \\ \text{Movement} \quad \text{Collision (or Interaction)} \\ \text{Kinematic} \quad \text{Dissipation} \\ C[f, f_2] = \text{Gain (scattered into) - Loss (scattered out)} \\ = \left(\frac{\delta f}{\delta t}\right)^+ - \left(\frac{\delta f}{\delta t}\right)^- \\ \sim \int |\mathbf{v} - \mathbf{v}_2| (f^* f_2^* - ff_2) d\mathbf{v}_2 \end{cases}$

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How complexity out of simplicity? - I

 $\left(\frac{\partial}{\partial t} + \mathbf{v} \cdot \nabla\right) f(t, \mathbf{r}, \mathbf{v}) = C[f, f_2] \qquad \qquad \rho \mathbf{u} = \langle m \mathbf{v} f(t, \mathbf{r}, \mathbf{v}) \rangle$ where $\langle \cdots \rangle = \iiint \cdots dv_x dv_y dv_z$

Differentiating the statistical definition $\rho \mathbf{u} = \langle m \mathbf{v} f(t, \mathbf{r}, \mathbf{v}) \rangle$ with time and then combining with BTE $(t, \mathbf{r}, \mathbf{v})$ are independent and $\mathbf{v} = \mathbf{u} + \mathbf{c}$)

$$\frac{\partial}{\partial t} \langle m\mathbf{v}f \rangle = \left\langle m\mathbf{v}\frac{\partial f}{\partial t} \right\rangle = -\left\langle m(\mathbf{v}\cdot\nabla f)\mathbf{v} \right\rangle + \left\langle m\mathbf{v}C[f,f_2] \right\rangle$$

$$[\mathbf{A}]^{(2)}: \text{ Traceless symmetric}$$
Here $-\left\langle m(\mathbf{v}\cdot\nabla f)\mathbf{v} \right\rangle = -\nabla \cdot \left\langle m\mathbf{v}\mathbf{v}f \right\rangle = -\nabla \cdot \left\{ \rho\mathbf{u}\mathbf{u} + \left\langle m\mathbf{c}\mathbf{c}f \right\rangle \right\}$

$$[\mathbf{A}]^{(2)}: \text{ Traceless symmetric}$$

$$part of \text{ tensor } \mathbf{A}$$

After the decomposition of the stress into pressure and viscous shear stress $\mathbf{\Pi}$ $\mathbf{P} \equiv \langle m\mathbf{cc}f \rangle = p\mathbf{I} + \mathbf{\Pi}$ where $p \equiv \langle m\mathbf{Tr}(\mathbf{cc})f/3 \rangle$, $\mathbf{\Pi} \equiv \langle m[\mathbf{cc}]^{(2)}f \rangle$,

and using the collisional invariance of the momentum, $\langle m\mathbf{v}C[f, f_2] \rangle = 0$, we have

$$\frac{\partial(\rho \mathbf{u})}{\partial t} + \nabla \cdot \left(\rho \mathbf{u} \mathbf{u} + p \mathbf{I} + \mathbf{I}\right) = \mathbf{0}.$$
 Exact consequence of the original BTE!

How complexity out of simplicity? - II

This procedure can be generalized for arbitrary molecular expressions of general moment (**viscous shear stress** Π and **heat flux Q**)

$$\rho \frac{D}{Dt} \begin{bmatrix} \mathbf{\Pi} \left(\equiv \left\langle m [\mathbf{cc}]^{(2)} f \right\rangle \right) / \rho \\ \mathbf{Q} \left(\equiv \left\langle mc^{2}\mathbf{c} / 2f \right\rangle \right) / \rho \end{bmatrix} + \nabla \cdot \begin{bmatrix} \left\langle m\mathbf{ccc}f \right\rangle - \left\langle m\mathrm{Tr}(\mathbf{ccc})f \right\rangle \mathbf{I} / 3 \\ \left\langle mc^{2}\mathbf{cc}f / 2 \right\rangle - C_{p}T(p\mathbf{I} + \mathbf{\Pi}) + \left\langle m\mathbf{ccc}f \right\rangle : \nabla \mathbf{u} \end{bmatrix} + \begin{bmatrix} 2[\mathbf{\Pi} \cdot \nabla \mathbf{u}]^{(2)} \\ \frac{D\mathbf{u}}{Dt} \cdot \mathbf{\Pi} + \mathbf{Q} \cdot \nabla \mathbf{u} + \mathbf{\Pi} \cdot C_{p}\nabla T \end{bmatrix} + \begin{bmatrix} 2p[\nabla \mathbf{u}]^{(2)} \\ pC_{p}\nabla T \end{bmatrix} = \begin{bmatrix} \left\langle m[\mathbf{cc}]^{(2)} C[f, f_{2}] \right\rangle \\ \left\langle (mc^{2} / 2 - mC_{p}T)\mathbf{c}C[f, f_{2}] \right\rangle \end{bmatrix}$$

2nd-order closure

Key points: 1) No approximation: exact consequence of the original BTE

2) Balanced closure for two open terms (Myong, 2014)

Different from previous practice misguided by Maxwellian molecule assumption

Do not cut the tiptoe in order to fit the foot into a shoe! (刖趾適屨)



How complexity out of simplicity? - III

H------ Hyperbolic (inviscid)

Conservation laws (exact consequence of BTE)

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$$\rho \frac{D}{Dt} \begin{bmatrix} 1/\rho \\ \mathbf{u} \\ E_t \end{bmatrix} + \nabla \cdot \begin{bmatrix} -\mathbf{u} \\ p\mathbf{I} \\ p\mathbf{u} \end{bmatrix} + \nabla \cdot \begin{bmatrix} 0 \\ (\mathbf{\Pi}) \\ \mathbf{\Pi} \cdot \mathbf{u} + \mathbf{Q} \end{bmatrix} = \mathbf{0}$$

in conjunction with the (complex) 2nd-order constitutive equations

$$\rho \frac{D(\Pi) \rho}{Dt} + 2[\Pi \cdot \nabla \mathbf{u}]^{(2)} + 2p[\nabla \mathbf{u}]^{(2)} = -\frac{p}{\mu_{NS}} \Pi q_{2nd}(\kappa_1),$$

$$\rho \frac{D(Q) \rho}{Dt} + \frac{D\mathbf{u}}{Dt} \cdot \Pi + \mathbf{Q} \cdot \nabla \mathbf{u} + \Pi \cdot C_p \nabla T + C_p p \nabla T = -\frac{pC_p}{k_{NS}} \mathbf{Q} q_{2nd}(\kappa_1),$$
Non-hyperbolic
mplicit
$$q_{2nd}(\kappa_1) = \frac{\sinh \kappa_1}{\kappa_1}, \quad \kappa_1 = \frac{T^{1/4}}{p} \left(\frac{\Pi : \Pi}{\mu_{NS}} + \frac{\mathbf{Q} \cdot \mathbf{Q} / T}{k_{NS}}\right)^{1/2}$$
Navier-Fourier laws inclusive
like onion!

A model for 2nd-order constitutive law

2nd-order constitutive laws beyond Navier law I



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2nd-order constitutive laws beyond Navier law II



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1-D shock result: inverse shock thickness



Shock density thickness of argon gas

Shock temperature-density distance

(Karchani, PhD Thesis, GNU, 2017)

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Multi-dimensional CFD based on decomposition

Acknowledgements (3rd Generation of Prof. Roe)

Former PhDs & Postdocs: A. Karchani (ANSYS, US), S. Singh (NTU, Singapore), H. Xiao (NWPU, China)

PhD Students: O. Ejtehadi, A. Rahimi, T. Chourushi

Supported by

National Research Foundation in Korea



Aerospace Computational Modeling Lab.

http://acml.gnu.ac.kr

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Comparison with 1st-order simple Burger

$$\frac{\partial u}{\partial t} + \frac{\partial}{\partial x} \left(\frac{1}{2}u^{2}\right) + \left|\frac{\partial \Pi}{\partial x}\right| = \text{ where } \Pi_{NS}\Pi + \Pi_{NS} = \Pi q(\Pi) \text{ and } \Pi_{NS} = -\eta \frac{\partial u}{\partial x}$$
New 2nd-order model
$$\begin{cases} \frac{\partial u}{\partial t} + \frac{\partial}{\partial x} \left(\frac{1}{2}u^{2} + \Pi\right) = 0, \\ \Pi = f^{-1}(-S) \text{ where } f(\Pi) = \frac{\sinh \Pi}{\Pi + 1}, \\ S = \eta \frac{\partial u}{\partial x} \end{cases}$$
Original 1st-order Burger
$$\begin{cases} \frac{\partial u}{\partial t} + \frac{\partial}{\partial x} \left(\frac{1}{2}u^{2} + \Pi\right) = 0, \\ \Pi = -S, \\ S = \eta \frac{\partial u}{\partial x} \end{cases}$$

 $\Pi_{NS}\Pi + \Pi_{NS} = \Pi q(\Pi) \text{ can be solved for given } \Pi_{NS} \text{ via method of iteration}$ $\Pi_{n+1} = \sinh^{-1}(\Pi_{NS}(1+\Pi_n)) \text{ for positive } \Pi_{NS},$ $\Pi_{n+1} = \frac{\Pi_{NS}}{q(\Pi_n) - \Pi_{NS}} \text{ for negative } \Pi_{NS}$

Decomposed computation in multi-dimension I

Edge based finite volume formulation in general coordinates

Decomposed computation in two-dimension II

$$\frac{\partial}{\partial t} \begin{pmatrix} \rho \\ \rho u \\ \rho v \\ \rho E \end{pmatrix} + \nabla \cdot \begin{pmatrix} \rho(\mathbf{u} \cdot \mathbf{n}) \\ \rho u(\mathbf{u} \cdot \mathbf{n}) + n_x p \\ \rho v(\mathbf{u} \cdot \mathbf{n}) + n_y p \\ \rho H(\mathbf{u} \cdot \mathbf{n}) \end{pmatrix} + \nabla \cdot \begin{pmatrix} 0 \\ n_x \Pi_{xx} + n_y \Pi_{xy} \\ n_x \Pi_{yx} + n_y \Pi_{yy} \\ n_x \Theta_x + n_y \Theta_y \end{pmatrix} = 0$$
where $\Theta_x = u \Pi_{xx} + v \Pi_{xy} + Q_x$, $\Theta_y = u \Pi_{yx} + v \Pi_{yy} + Q_y$

In the case of *x*-direction, $(\Pi_{xx}, \Pi_{xy}, Q_x)$ on interface for a given input (u_x, v_x, T_x) can be **decomposed** as

 $\begin{aligned} f_{x}(u_{x},v_{x},T_{x}) &= f_{1x}(u_{x},0,T_{x}) + f_{2x}(0,v_{x},0).\\ \text{Similarly,} \quad f_{y}(u_{y},v_{y},T_{y}) &= f_{1y}(0,v_{y},T_{y}) + f_{2y}(u_{y},0,0).\\ \text{Finally we have} \\ \Pi_{xx} &= \Pi_{xx-1x} + \Pi_{xx-2x} + \Pi_{xx-1y} + \Pi_{xx-2y}, \qquad 2[\Pi \cdot \nabla \mathbf{u}] \\ \Pi_{xy} &= \Pi_{xy-2x} + \Pi_{xy-2y}, \qquad \frac{D\mathbf{u}}{Dt} \cdot \Pi + \\ \Pi_{yy} &= \Pi_{yy-1x} + \Pi_{yy-2x} + \Pi_{yy-1y} + \Pi_{yy-2y}, \qquad Q_{x} = Q_{x-1x}, \quad Q_{y} = Q_{y-1y}, \qquad Q_{2nd}(\kappa_{1}) = 0 \end{aligned}$

$$2\left[\boldsymbol{\Pi} \cdot \nabla \boldsymbol{u}\right]^{(2)} + 2p\left[\nabla \boldsymbol{u}\right]^{(2)} = -\frac{p}{\mu_{NS}} \boldsymbol{\Pi} q_{2nd}(\kappa_1),$$

$$\frac{D\boldsymbol{u}}{Dt} \cdot \boldsymbol{\Pi} + \boldsymbol{Q} \cdot \nabla \boldsymbol{u} + \boldsymbol{\Pi} \cdot \boldsymbol{C}_p \nabla T + \boldsymbol{C}_p p \nabla T = -\frac{p\boldsymbol{C}_p}{k_{NS}} \boldsymbol{Q} q_{2nd}(\kappa_1)$$

$$q_{2nd}(\kappa_1) \equiv \frac{\sinh \kappa_1}{\kappa_1}, \ \kappa_1 \equiv \frac{T^{1/4}}{p} \left(\frac{\boldsymbol{\Pi} : \boldsymbol{\Pi}}{\mu_{NS}} + \frac{\boldsymbol{Q} \cdot \boldsymbol{Q}/T}{k_{NS}}\right)^{1/2}$$

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Mixed modal DG method for the 2nd-order model I

$$\partial_t \mathbf{U} + \nabla \mathbf{F}_{inv}(\mathbf{U}) + \nabla \mathbf{F}_{vis}(\mathbf{U}, \nabla \mathbf{U}) = 0$$

Discretization in mixed form

$$\begin{cases} \mathbf{S} - \nabla \mathbf{U} = \mathbf{0} \\ \partial_t \mathbf{U} + \nabla \mathbf{F}_{inv}(\mathbf{U}) + \nabla \mathbf{F}_{vis}(\mathbf{U}, \mathbf{S}) = \mathbf{0} \end{cases}$$

NSF model (
$$\Pi$$
, \mathbf{Q}) = $\mathbf{f}_{\text{linear}}(\mathbf{S}(\mathbf{U}))$
NCCR model (Π , \mathbf{Q})_{NCCR} = $\mathbf{f}_{\text{non-linear}}(\mathbf{S}(\mathbf{U}), p, T)$
 $\mathbf{U}_{h}(\mathbf{x},t) = \sum_{i=0}^{k} U_{j}^{i}(t) \varphi^{i}(\mathbf{x}), \quad \mathbf{S}_{h}(\mathbf{x},t) = \sum_{i=0}^{k} S_{j}^{i}(t) \varphi^{i}(\mathbf{x})$
NCCR: Nonlinear Coupled
Constitutive Relation

$$\begin{cases} \frac{\partial}{\partial t} \int_{I} \mathbf{U} \varphi dV - \int_{I} \nabla \varphi \mathbf{F}_{\text{inv}} dV + \int_{\partial I} \varphi \mathbf{F}_{\text{inv}} \cdot \mathbf{n} d\Gamma - \int_{I} \nabla \varphi \mathbf{F}_{\text{vis}} dV + \int_{\partial I} \varphi \mathbf{F}_{\text{vis}} \cdot \mathbf{n} d\Gamma = 0, \\ \int_{I} \mathbf{S} \varphi dV + \int_{I} T^{s} \nabla \varphi \mathbf{U} dV - \int_{\partial I} T^{s} \varphi \mathbf{U} \cdot \mathbf{n} d\Gamma = 0, \end{cases}$$

Dubiner basis function, Lax-Friedrichs inviscid flux, central flux for viscous terms

Mixed modal DG method for the 2nd-order model II

Convergence property and super-parallel performance

$$\hat{\Pi}\hat{\Pi}_{NS} + \hat{\Pi}_{NS} = \hat{\Pi}q(\hat{\Pi})$$

Triangular mesh partition using ParMETIS.; different colors represent subdomains owned by different processors.



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Two-dimensional shock structure around cylinder



Computed Mach contours over cylinder (Mach 5.48, Kn=0.2)

NCCR: Nonlinear Coupled Constitutive Relation

(Karchani, PhD Thesis, GNU, 2017)

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Three-dimensional rarefied hypersonic flows

A suborbital re-entry vehicle



First-order NSF solution

Second-order NCCR solution NCCR: Nonlinear Coupled Constitutive Relation

Mach contours of gas flows; M = 5.0, Kn = 0.02