

Modeling of Gaseous Expansion in Free Jet and Microchannel Flows Using the Modified Moment Method

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Talk outline

- Research goals
- Derivation of hydrodynamic equations from the Boltzmann equation:
the moment method
- Gaseous expansion in free jet
- Molecular interaction model for gas particle and surface atoms
- Gaseous expansion in micro-channel
- Summary

Research goals

- A fundamental question:
Can the moment method based on the reduction of information be capable of describing high non-equilibrium gas flows (rarefied and microscale)?
- Research goals:
Answering fundamental questions and validation study
Deriving governing equation and boundary conditions
 -> the basic building block for an **efficient** CFD code
Providing physical insight and economy of thoughts
- Practical issues:
How to solve the moment equations?

Previous works

- Linear theory: Navier-Stokes

Suitable for preliminary calculation, very efficient and powerful (modern CFD codes), question of applicability

- Molecular description in phase space: Boltzmann equation, DSMC, MD, etc

Valid for whole flow regimes, non-trivial issue in computational efficiency

- High order hydrodynamic theories in thermodynamic space:

Chapman-Enskog method (Burnett, super-Burnett, BGK-Burnnet), Grad's moment method, (rational) extended irreversible thermodynamics, information entropy maximization method...

Achieving economy of thoughts and description, problem in non-physical solutions and defining the boundary quantities

Strategy

- **Capability:** recover the Navier-Stokes model and extend to the transition regimes [Kn and $Kn \cdot M \sim O(1)$]
- **Constraints** on the new model: efficiency, robustness, and mathematical consistency
- **Issues:**
 - Equation types**—hyperbolic or parabolic \implies no priori restriction
 - Boundary conditions**—only conserved quantities as independent variables \implies no definite information of non-conserved variables on the boundary available
 - Balanced treatment** of each effects (convective, kinematic, and dissipative) \implies dimensionless parameters as a guide
- **Consider a simple problem (benchmark problem)**
 - Focus on the big picture**

The benchmark problems

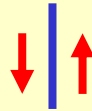
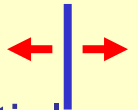
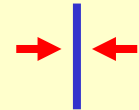
- **Why benchmark problems?:** Simple but preserve the essence of physics; experimental study possible

- 3 cases:

(-) **normal velocity gradient**—compressive nature—shock wave

(+) **normal velocity gradient**—expansive nature—free jet

Shear velocity gradient—shear-driven velocity profile—tangential flow near the solid wall



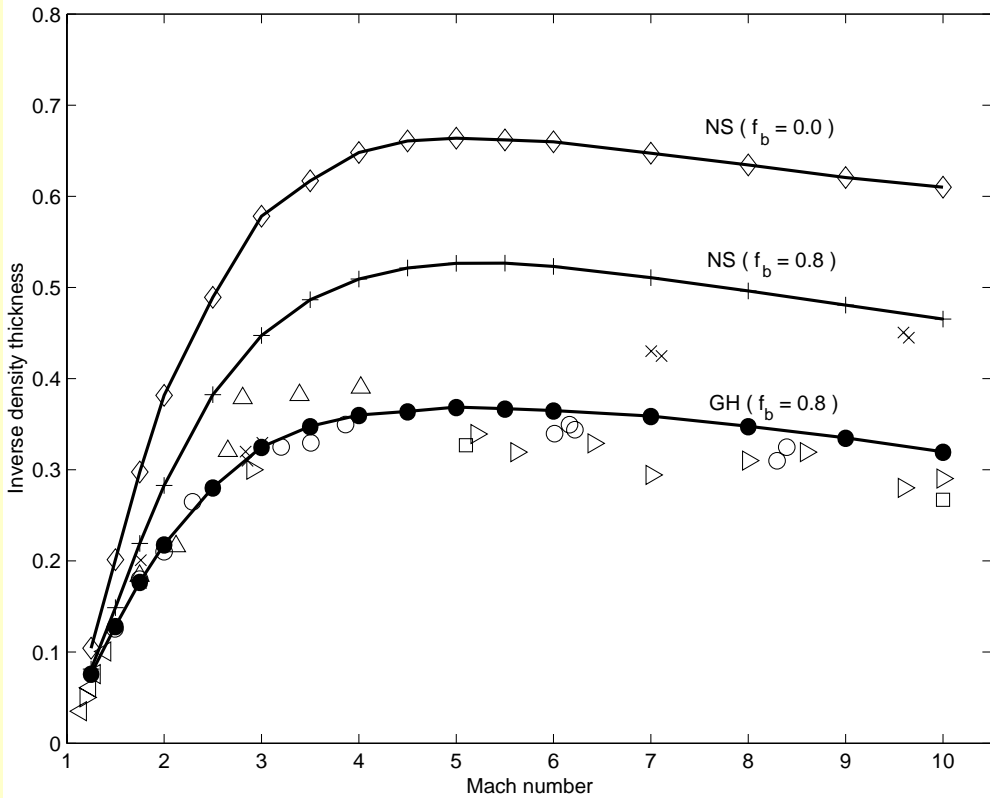
- **Shock wave:**

The most critical problem since the Boltzmann collision integral is directly involved

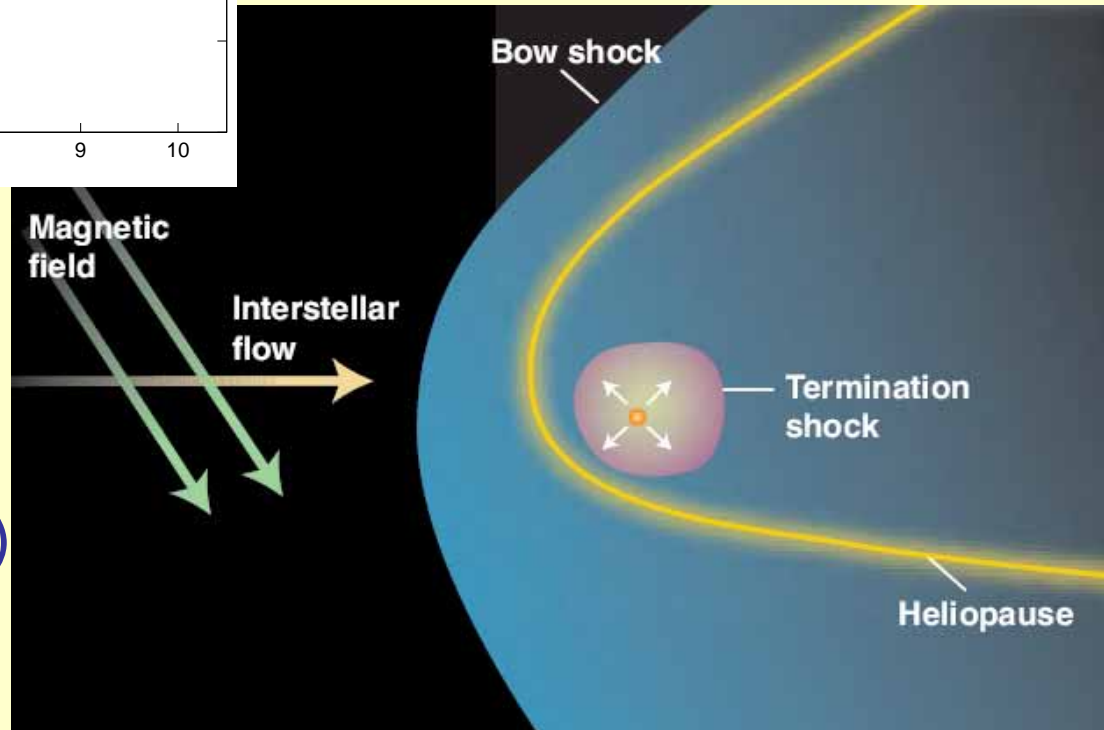
Without the usual complication from the boundary

Shock wave examples

- hypersonic shock wave over a blunt body



- Interaction of the solar wind with interstellar medium (plasma and neutral hydrogen gas)



The moment method (Maxwell and Grad)

- Derivation of the conservation laws and constitutive equations using the concept of statistical average

By differentiating **stress tensor** $\mathbf{P} = \langle m\mathbf{c}\mathbf{c}f(\mathbf{v}, \mathbf{r}, t) \rangle$ with time and combining with the Boltzmann equation $(\partial_t + \mathbf{v} \cdot \nabla)f(\mathbf{v}, \mathbf{r}, t) = C[f]$,

we obtain $\mathbf{P}_t = -\langle m\mathbf{c}\mathbf{c}(\mathbf{v} \cdot \nabla f) \rangle + \Lambda^{(P)}$ where $\Lambda^{(P)} \equiv \langle m\mathbf{c}\mathbf{c}C[f] \rangle$. ($\mathbf{P} \equiv p\mathbf{I} + \mathbf{\Pi}$)

$$\rho \frac{D}{Dt} \left(\frac{\mathbf{\Pi}}{\rho} \right) + \nabla \cdot \boldsymbol{\Psi}^{(\Pi)} = -2[\mathbf{\Pi} \cdot \nabla \mathbf{u}]^{(2)} - 2p[\nabla \mathbf{u}]^{(2)} + \Lambda^{(\Pi)} \left(\equiv \langle m[\mathbf{c}\mathbf{c}]^{(2)} C[f] \rangle \right)$$

substantial derivative of non - conserved variable

+ flux of high order flux = kinematic term + **dissipation term (Boltzmann collision)**

Mathematically, $\mathbf{\Pi}$ = function of conserved variables and their gradients

- ➡ No approximations; main parameter $\Pi / p \sim \text{Kn} \cdot M$ (not Kn alone)
 No explicit $C[f(\mathbf{r}, \mathbf{v}, t)]$ except for the dissipation term; not pure hyperbolic

$$\rho \frac{D}{Dt} \begin{bmatrix} [1, \rho \mathbf{u}, \rho E_t]^T / \rho \\ \mathbf{\Pi} / \rho \\ \mathbf{Q} / \rho \end{bmatrix} + \nabla \cdot \begin{bmatrix} [\mathbf{u}, \mathbf{P}, \mathbf{P} \cdot \mathbf{u} + \mathbf{Q}]^T \\ \boldsymbol{\Psi}^{(\Pi)} \\ \boldsymbol{\Psi}^{(Q)} \end{bmatrix} = \begin{bmatrix} \mathbf{0} \\ \mathbf{Z}^{(\Pi)} + \Lambda^{(\Pi)} \\ \mathbf{Z}^{(Q)} + \Lambda^{(Q)} \end{bmatrix}$$

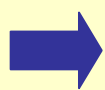
Eu's modified moment method

- **Grad's moment method (1949)**

$$\rho \frac{D}{Dt} \left(\frac{\mathbf{\Pi}}{\rho} \right) + \nabla \cdot \boldsymbol{\Psi}^{(\Pi)} = -2[\mathbf{\Pi} \cdot \nabla \mathbf{u}]^{(2)} - 2p[\nabla \mathbf{u}]^{(2)} + \Lambda^{(\Pi)} \left(\equiv \left\langle m[\mathbf{c}\mathbf{c}]^{(2)} C[f] \right\rangle^{Grad} = - \frac{\mathbf{\Pi}}{\eta/p} \right)$$

Relaxation (BGK) approximation for $C[f]$ and f in a polynomial form

Closure relation: high order moment $\boldsymbol{\Psi}^{(\Pi)} \sim$ heat flux



Mathematical singularity for high $Kn * M$

Difficulty in defining moments (stress) at the boundary

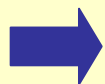
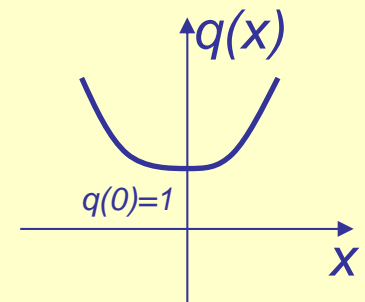
- **Eu's modified moment method (1980, 1992, 1998, 2002)**

$$\rho \frac{D}{Dt} \left(\frac{\mathbf{\Pi}}{\rho} \right) + \nabla \cdot \boldsymbol{\Psi}^{(\Pi)} = -2[\mathbf{\Pi} \cdot \nabla \mathbf{u}]^{(2)} - 2p[\nabla \mathbf{u}]^{(2)} + \Lambda^{(\Pi)} \left(\equiv \left\langle m[\mathbf{c}\mathbf{c}]^{(2)} C[f] \right\rangle^{Eu} = - \frac{\mathbf{\Pi}}{\eta/p} q(\kappa) \right)$$

f in an exponential form

Cumulant expansion for $C[f]$

$$q(\kappa) \equiv \frac{\sinh \kappa}{\kappa} \text{ where } \kappa = \frac{(mk_B T)^{1/4}}{\sqrt{2}pd} \left(\frac{\mathbf{\Pi} : \mathbf{\Pi}}{2\eta} + \frac{\mathbf{Q} \cdot \mathbf{Q}}{\lambda} \right)^{1/2}$$

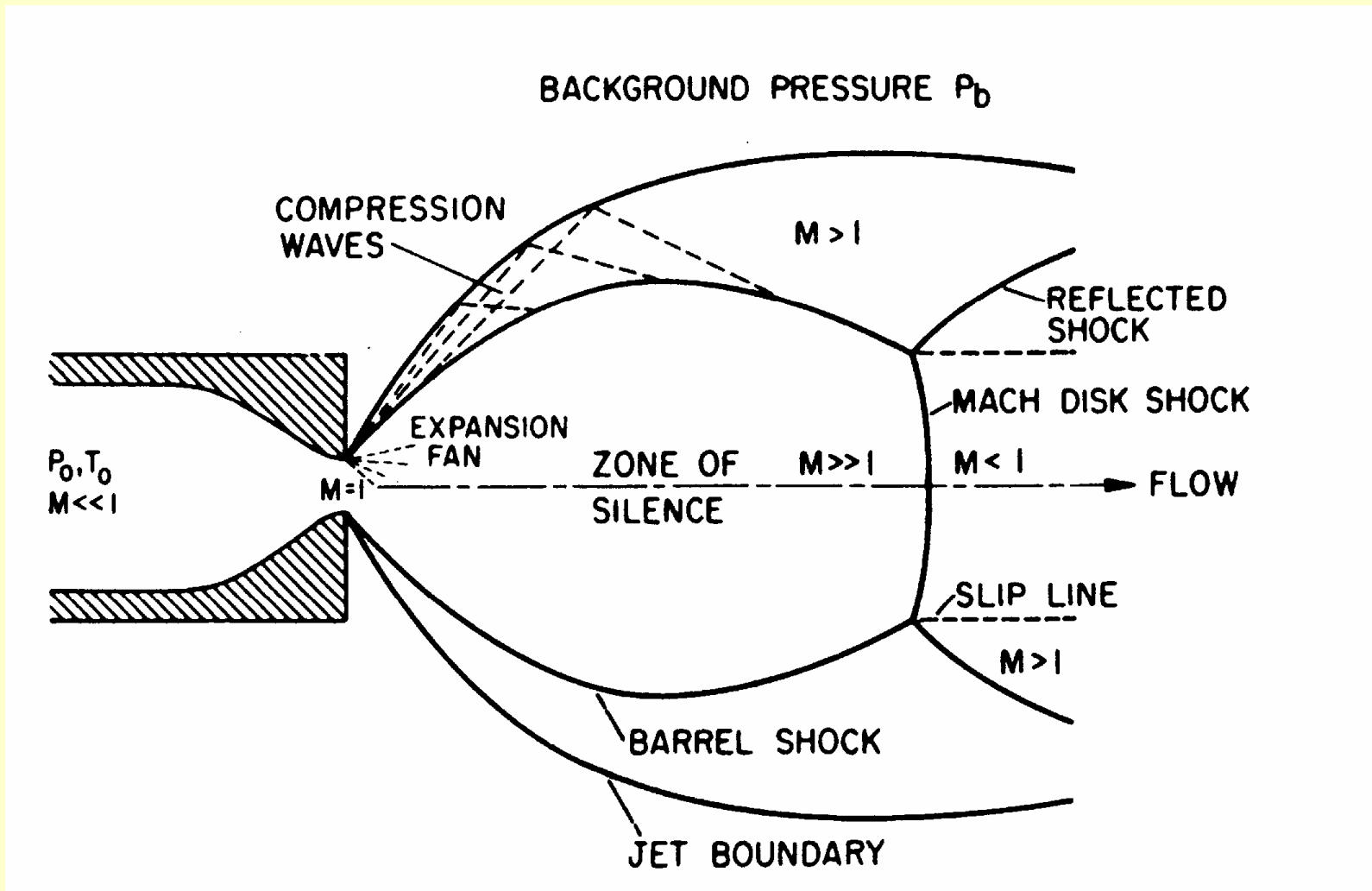


Mathematical singularity is removed

Differential \rightarrow algebraic equations \rightarrow resolve the boundary problem

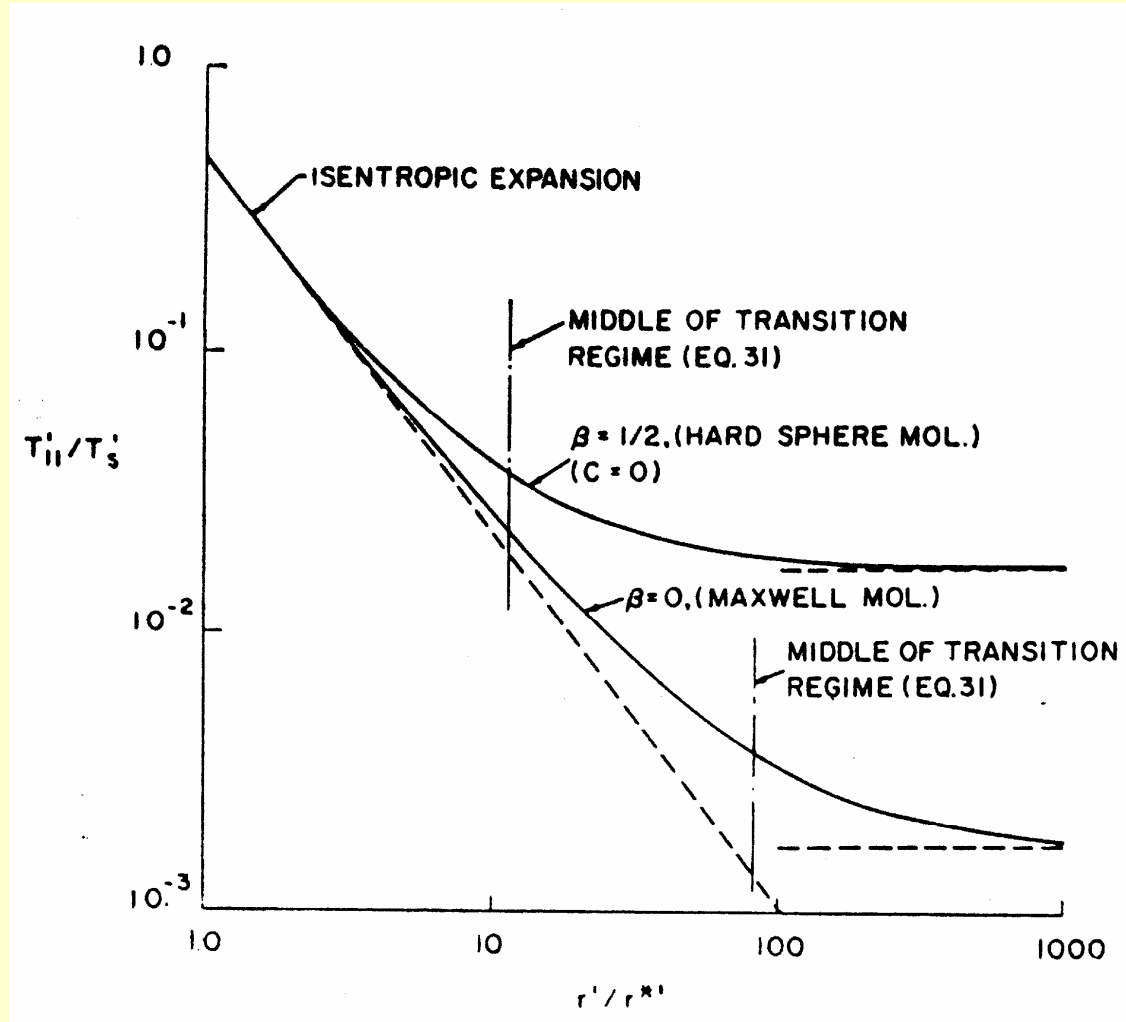
Structure of the free jet

- Practical need of production of high-intensity molecular beam
- Axisymmetric flow configuration: spherically symmetric as an approximation



Pure spherically symmetric source flow

- Kinetic theory of source flow expansion by Hamel and Willis (1966)
- Unusual behavior: freezing of the parallel temperature ($p + \Pi_{rr}$)



Analysis by the modified moment method (I)

- Radial dependency only (ρ, u, p, Π_{rr}, Q)

$$\frac{d}{dr} \begin{bmatrix} \rho u \\ \rho u^2 + p \\ \rho(h + u^2/2) \end{bmatrix} + \frac{2}{r} \rho u \begin{bmatrix} 1 \\ u \\ h + u^2/2 \end{bmatrix} + \left[\begin{array}{c} 0 \\ \frac{1}{r^3} \frac{d}{dr} (r^3 \Pi_{rr}) \\ \Pi_{rr} \left(\frac{du}{dr} - \frac{u}{r} \right) + \frac{u}{r^3} \frac{d}{dr} (r^3 \Pi_{rr}) + \frac{1}{r^2} \frac{d}{dr} (r^2 Q) \end{array} \right] = 0$$

$$\Pi_{rr} q(\kappa) = \Pi_{rr0} \left[1 + \frac{\Pi_{rr}}{p} \left(1 + \frac{3}{2} \frac{u/r}{du/dr - u/r} \right) \right]$$

dissipation term

kinematic terms

geometrical effect (spherical symmetry)

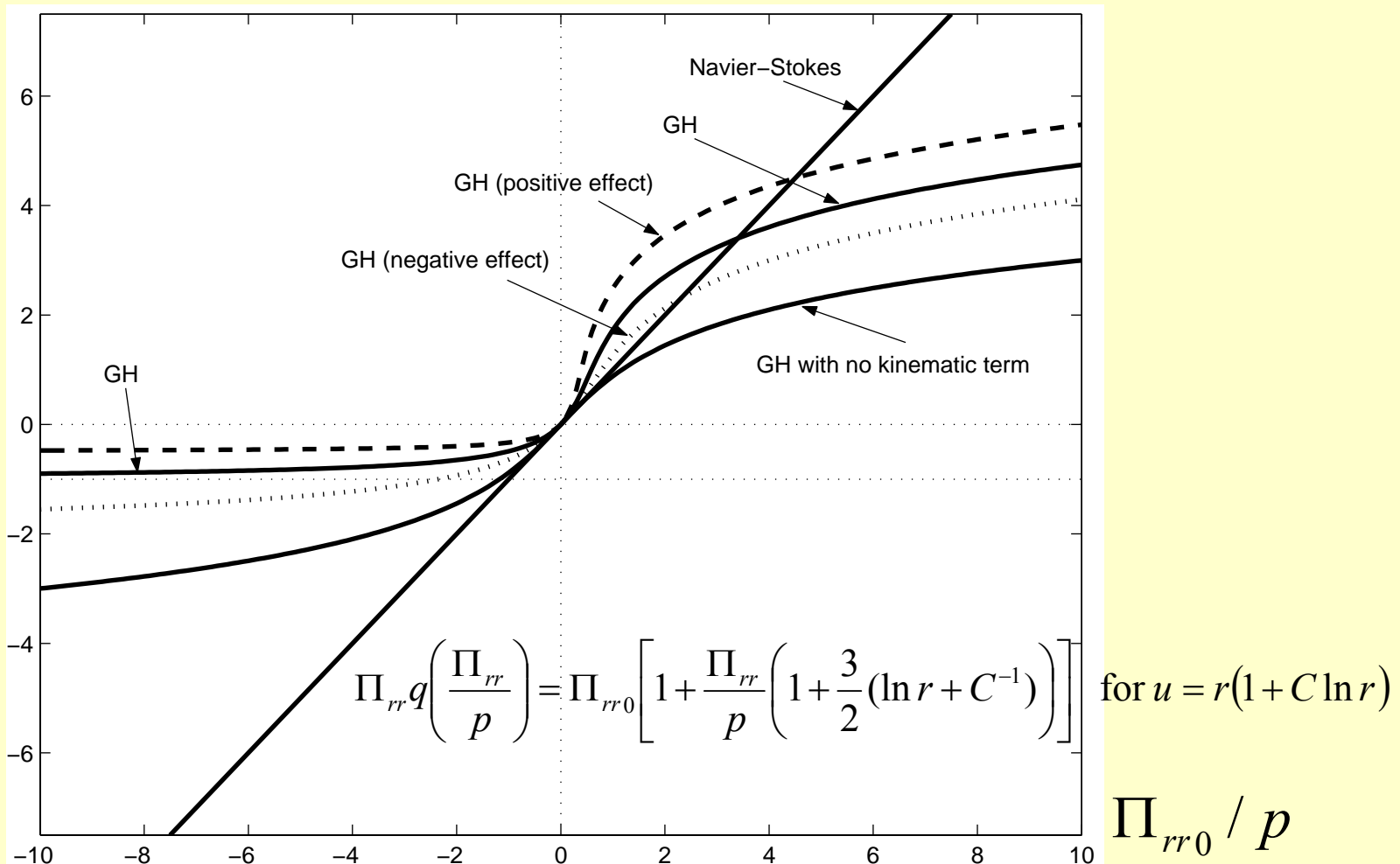
$$Qq(\kappa) = Q_0 \left[1 + \frac{\Pi_{rr}}{p} \right] \quad \text{where} \quad \kappa = \frac{(mk_B T)^{1/4}}{p\sqrt{2d}} \left[\frac{3/2\Pi_{rr}^2}{2\eta} + \frac{Q^2}{\lambda} \right]^{1/2}$$

$$\Pi_{rr0} = -\frac{4\eta}{3} r \frac{d}{dr} \left(\frac{u}{r} \right), \quad Q_0 = -\lambda \frac{d}{dr} (\ln T)$$

Analysis by the modified moment method (II)

- $p + \Pi_{rr}$ decreases in expanding flow as the radius increases -> a possible explanation of freezing of the parallel temperature $p + \Pi_{rr}$

Π_{rr} / p



Boundary condition based on Langmuir adsorption isotherm

N : number of sites (s) per unit area of the surface interacting with gas molecules (m)

$N\alpha$: number of sites which are covered

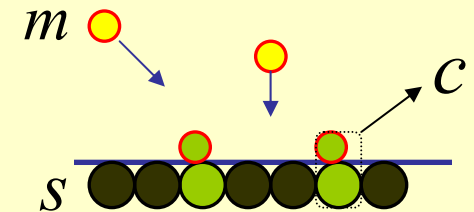
$N(1 - \alpha)$: number of sites which are not covered

Let us assume that m and s form the complex c . Then the equilibrium constant K becomes

$$K = \frac{C_c}{C_m C_s} = \frac{N\alpha}{[p / k_B T_w] N(1 - \alpha)},$$

that is,

$$\alpha = \frac{\beta p}{1 + \beta p} \quad \text{where} \quad \beta = \frac{K}{k_B T_w}.$$



Slip Boundary Conditions

- **Langmuir slip condition (Dirichlet type)**

$$u = \alpha u_w + (1 - \alpha)u_r, \quad T = \alpha T_w + (1 - \alpha)T_r \quad \text{where } \alpha = \frac{p / 4\omega Kn}{1 + p / 4\omega Kn}$$

$$\omega = \omega_0(\nu) \left(\frac{T_w}{T_r} \right)^{1+2/(\nu-1)} \exp\left(-\frac{D_e}{k_B T_w} \right) = \underline{fn(\nu, T_w, D_e)}$$

D_e : Heat of adsorption [$O(10^{-1} \sim 10)$ kcal/mol]

- **Maxwell slip condition (Neumann type)**

$$u = u_w + \boxed{\sigma_{\Pi_w}} = u_w + \sigma_v \ell \left(\frac{\partial u}{\partial n} \right)_w, \quad T = T_w + \sigma_T \frac{1}{Pr} \frac{2\gamma}{\gamma + 1} \ell \left(\frac{\partial T}{\partial n} \right)_w$$

Cf. We can prove $\omega \sim \sigma_v (\equiv (2 - \theta) / \theta)$ in the case of microchannel flow.

As a result, **a physical meaning can be assigned to $\sigma_{v,T}$.**

$$\sigma_{v,T} \sim \omega = \omega_0(\nu) \left(\frac{T_w}{T_r} \right)^{1+2/(\nu-1)} \exp\left(-\frac{D_e}{k_B T_w} \right)$$

Analysis and validation study

- Pressure-driven compressible flow in microchannel with finite length

$$\frac{\partial(pu)}{\partial x} + \frac{\partial(pv)}{\partial y} = 0$$

with slip b. c. \Rightarrow

$$p(x), u(x, y), v(x, y)$$

$$\text{Ex. } \frac{u(y)}{u_{ave}} = \frac{1/4 - y^2 + Kn}{1/6 + Kn}$$

$$\frac{dp}{dx} = \frac{\partial^2 u}{\partial y^2}$$

For high Kn^*M
flow \Rightarrow

$$\frac{dp}{dx} = -\frac{\partial \Pi_{xy}}{\partial y} \quad \text{with} \quad \Pi_{xy} q(CN_\delta |\Pi_{xy}| / p) = -\frac{\partial u}{\partial y}$$

$$\text{Ex. } \frac{u(y)}{u_{ave}} = \frac{1 - \cosh(\Phi / 2) - \alpha[1 - \cosh(\Phi y)]}{1 - \cosh(\Phi / 2) - \alpha[1 - q(\Phi / 2)]}$$

$$\text{where } \Phi = -C\varepsilon \frac{d(\ln p)}{dx}$$

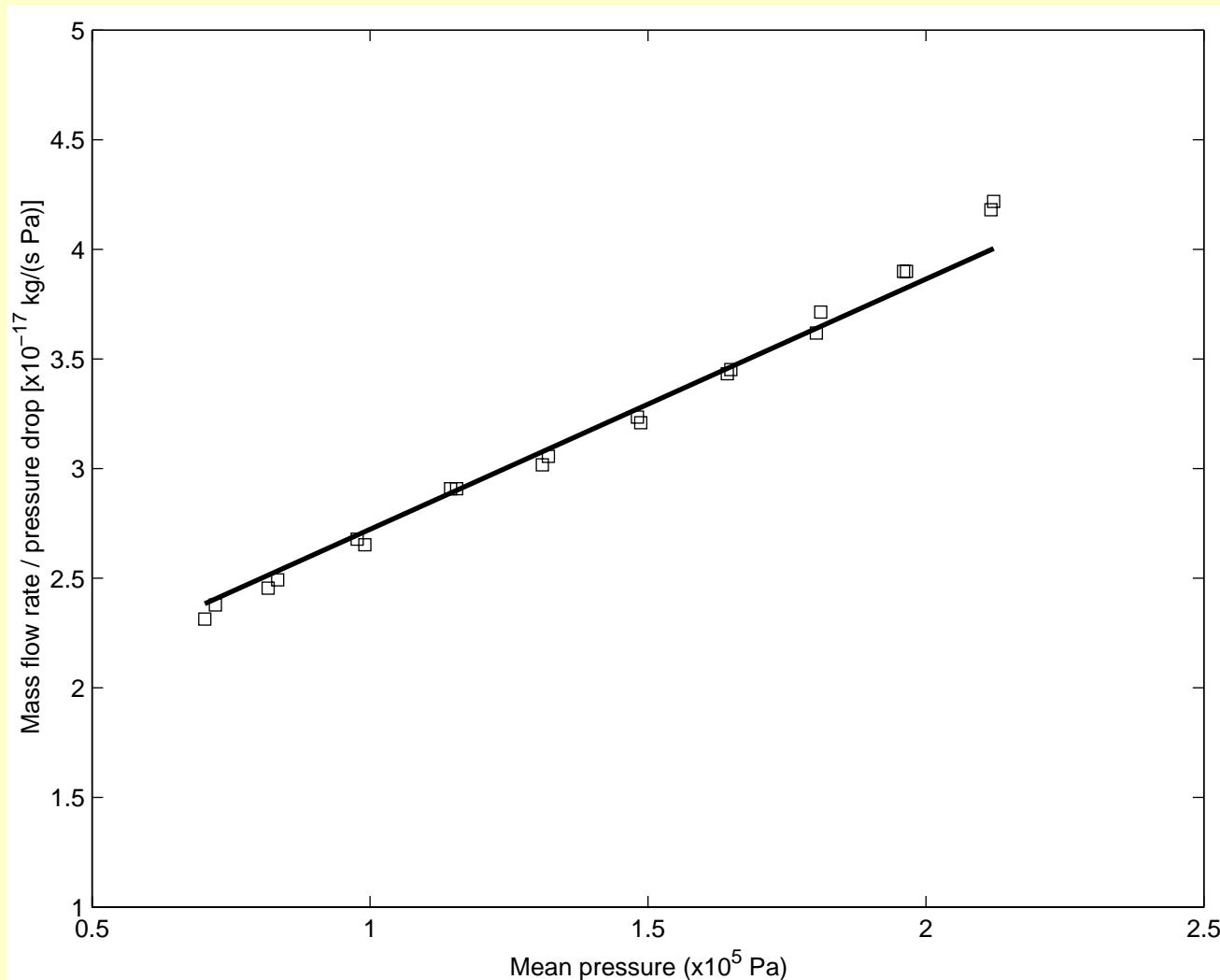
Numerical solutions for o. d. e. of pressure distribution

Ex. Arkilic's experiment in micro - channel

(silicon, $H = 1.33 \mu m$, $W = 52.3 \mu m$, $L = 7.5 mm$)

Helium gas (exit $Kn = 2.5$)

Mass flow rate of helium gas exhausting to an exit pressure of 6.5 kPa (Kn=2.5) (Arkilic's 1997 data)



Summary

- The modified moment method is applied to the study of benchmark problems; pressure-driven free jet and microchannel flows.
- The kinetic term associated with the ratio of radial stress components in the spherical geometry seems to be responsible for the freezing of the parallel temperature.
- The full results of the spherically symmetric source flow remains to be seen.